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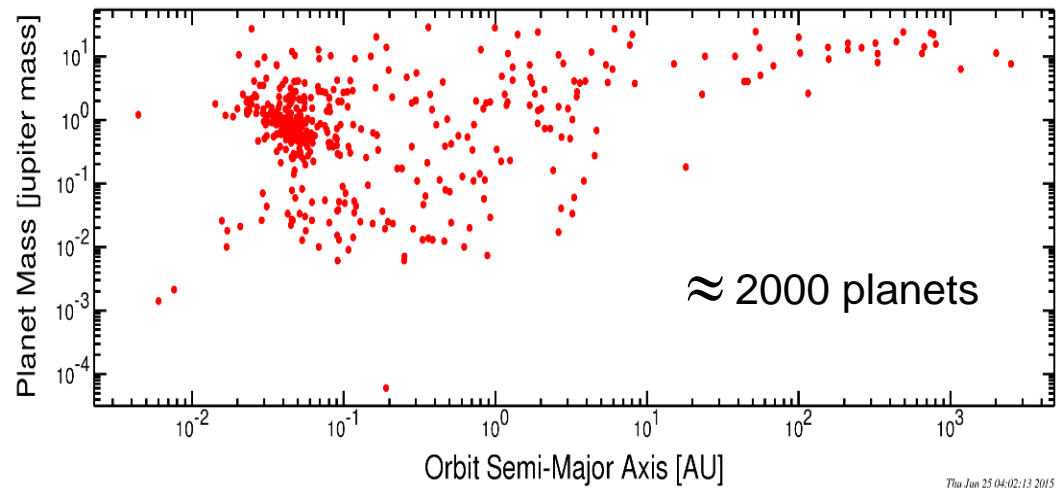


Siegfried Eggl
IMCCE, Observatoire de Paris, France

Analytic Orbit Propagation for Transiting Circumbinary Planets

12th HELLENIC ASTRONOMICAL CONFERENCE
Thessaloniki, June 28 – July 2, 2015

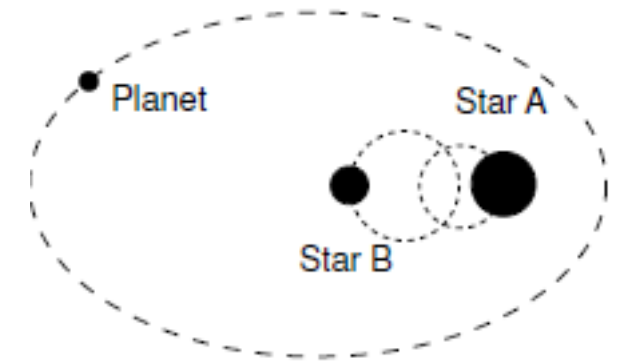
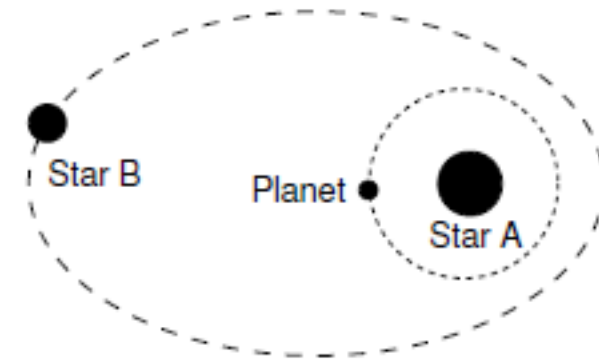
Confirmed Planets



140 planets in
binaries

114 S-type
(circumstellar)

26 P-type
(circumbinary)



Confirmed planets: 1028

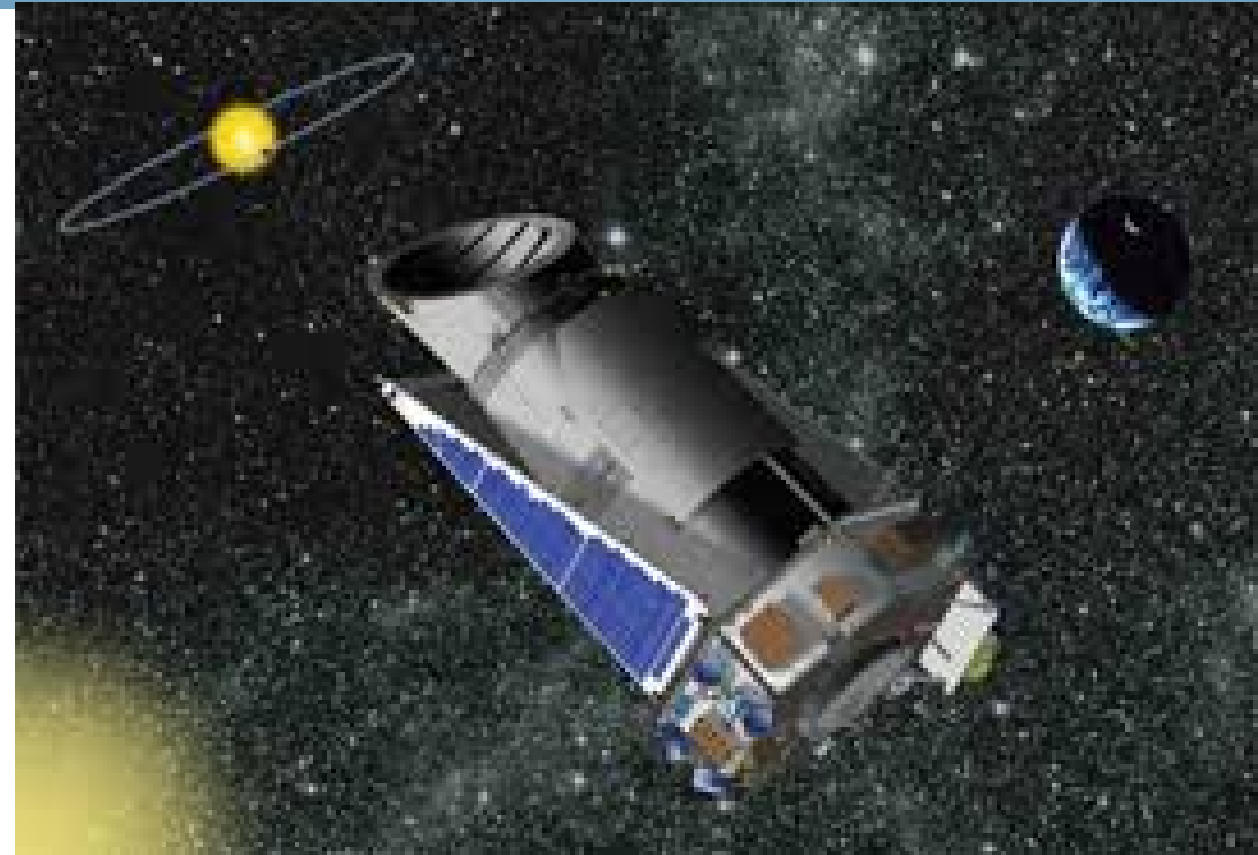
Planet candidates: 4664

Eclipsing binaries: 2165

Transit method



www.eso.org



P-type:

Kepler-16b, Kepler-34b, Kepler-35b, Kepler-38b, Kepler-47b,
Kepler-47c, Kepler-47d, Kepler-64b, Kepler-413b, KIC 9632895

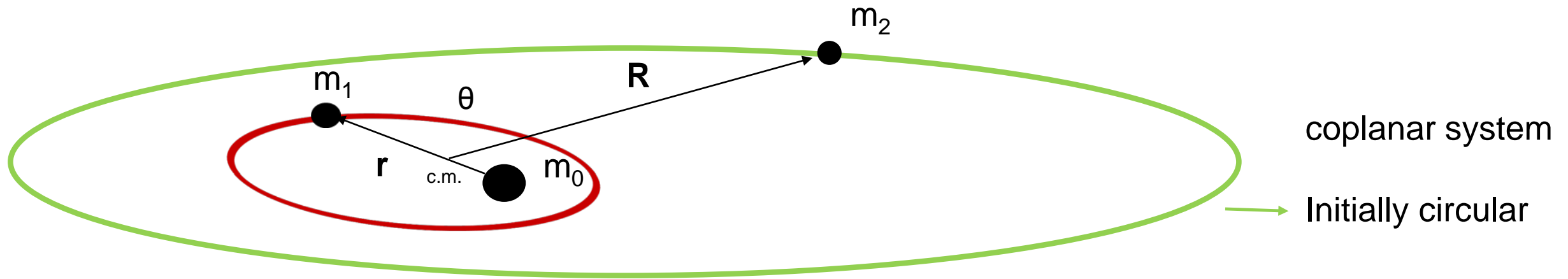
AIM: To provide an analytical description of a circumbinary planetary orbit

- modelling TTVs
- habitable zones in P-type systems (Eggl, Georgakarakos & Pilat-Lohinger 2015; in prep)
- low mass binaries
- planet formation in binaries

Hierarchical Triple System (HTS)

r , R Jacobi vectors

$$m_2 \ll m_0, m_1$$



Hierarchical Triple System (HTS)

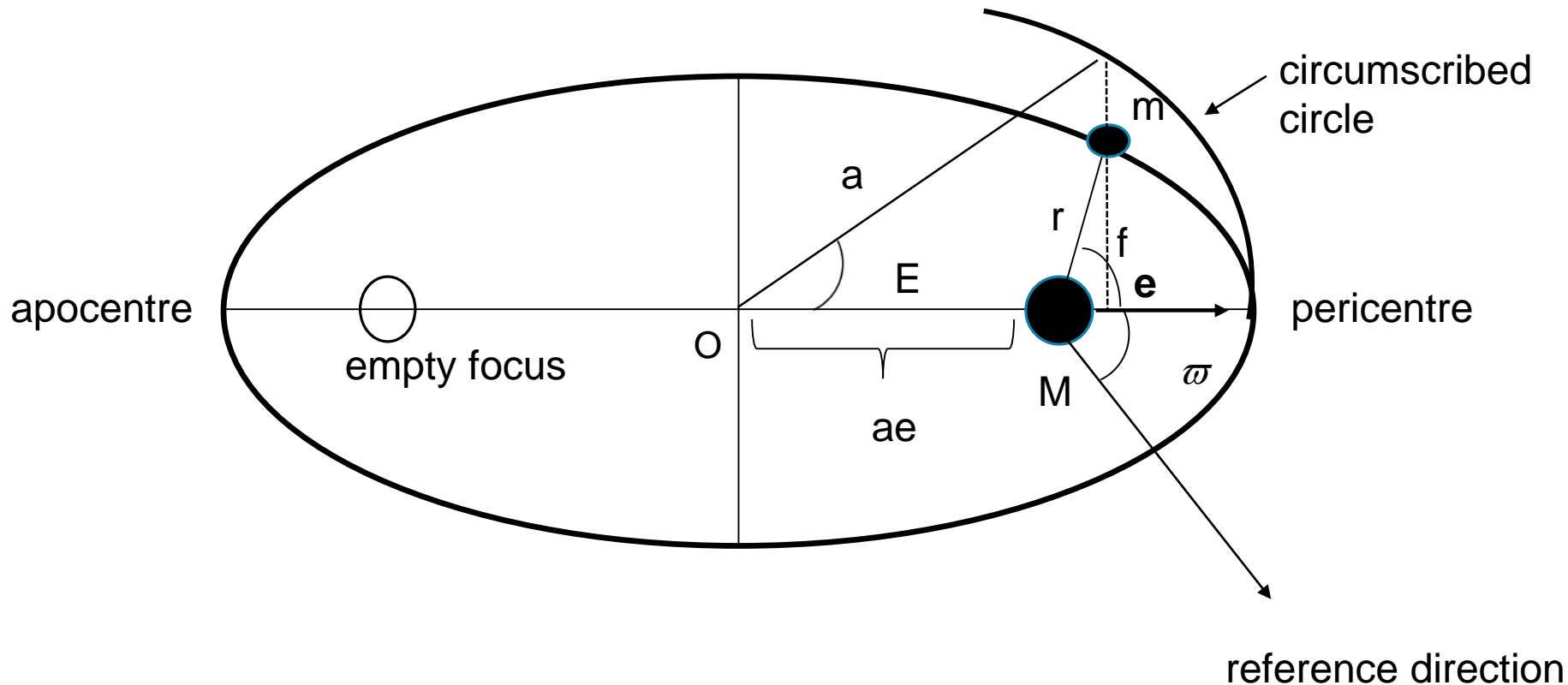
- No long term changes in the semi-major axes of the orbits (e.g. Harrington 1968).



+ coplanarity

the dynamical evolution of the hierarchical triple is dominated by the time dependent changes in the eccentric (Runge-Lenz) vectors of the inner and outer orbit

Two body problem



a = semi-major axis,
 e = eccentricity,
 \mathbf{e} = Runge-Lenz (eccentric) vector,
 ϖ = longitude of pericentre,
 E = eccentric anomaly,
 f = true anomaly
 T = period of the orbit,
 n = mean motion ($T=2\pi/n$),
 $l=nt+c$ mean anomaly
(c is a constant).

METHOD:

- long term (secular) evolution: Hamiltonian formulation
- mid term ($\sim T_2$) and short term ($\sim T_1$) evolution: Runge-Lenz vector
- no mean motion resonances ($n_1 \neq kn_2$, $k \in \mathbb{Z}$)
- semi-major axes and binary eccentricity constant

LONG TERM MOTION

$$H = H_0 + H_1 + H_p$$

r/R small

where

$$H_0 = -\frac{\mathcal{G}^2 m_0^3 m_1^3}{2(m_0 + m_1)L_1^2}$$

is the Keplerian energy of the inner orbit,

$$H_1 = -\frac{\mathcal{G}^2 (m_0 + m_1)^3 m_2^3}{2ML_2^2}$$

is the Keplerian energy of the outer orbit, and

$$H_p = \mathcal{G}m_2 \left(\frac{m_0 + m_1}{R} - \frac{m_0}{r_{02}} - \frac{m_1}{r_{12}} \right)$$

is the perturbing Hamiltonian, with r_{02} and r_{12} being the distances between m_0 and m_2 and m_1 and m_2 respectively

$$H_p = -\frac{\mathcal{G}m_0m_1m_2}{R} \sum_{j=2}^{\infty} M_j \left(\frac{r}{R}\right)^j \mathcal{P}_j(\cos \theta)$$

$$P_n(x) = \frac{(-1)^n}{2^n n!} \frac{d^n}{dx^n} (1 - x^2)^n$$

MID AND SHORT TERM EVOLUTION

Planetary equation of motion: $\ddot{\mathbf{R}} = -\mathcal{G}M \left(\mu_0 \frac{\mathbf{R} + \mu_1 \mathbf{r}}{|\mathbf{R} + \mu_1 \mathbf{r}|^3} + \mu_1 \frac{\mathbf{R} - \mu_0 \mathbf{r}}{|\mathbf{R} - \mu_0 \mathbf{r}|^3} \right) = \mathcal{G}M \frac{\partial}{\partial \mathbf{R}} \left(\frac{\mu_0}{|\mathbf{R} + \mu_1 \mathbf{r}|} + \frac{\mu_1}{|\mathbf{R} - \mu_0 \mathbf{r}|} \right)$ with

$$\mu_i = \frac{m_i}{m_0 + m_1}, \quad i = 0, 1$$

As previously, HTS - r/R small - use Legendre polynomials

$$\frac{1}{|\mathbf{R} + \mu_1 \mathbf{r}|} = \frac{1}{R} \sum_{n=0}^{\infty} \left(-\frac{\mu_1 r}{R} \right)^n \mathcal{P}_n(\cos \theta),$$

and

$$\frac{1}{|\mathbf{R} - \mu_0 \mathbf{r}|} = \frac{1}{R} \sum_{n=0}^{\infty} \left(\frac{\mu_0 r}{R} \right)^n \mathcal{P}_n(\cos \theta).$$

With $\cos \theta = \frac{\mathbf{r} \cdot \mathbf{R}}{rR}$

Runge-Lenz vector:

$$\begin{aligned} \mathbf{e}_2 &= -\frac{\mathbf{R}}{R} + \frac{1}{\mathcal{G}M} (\dot{\mathbf{R}} \times \mathbf{h}) \Rightarrow \\ \dot{\mathbf{e}}_2 &= -\frac{\dot{\mathbf{R}}}{R} + \frac{\dot{R}}{R^2} \mathbf{R} + \frac{1}{\mathcal{G}M} [2(\dot{\mathbf{R}} \cdot \ddot{\mathbf{R}}) \mathbf{R} - (\mathbf{R} \cdot \ddot{\mathbf{R}}) \dot{\mathbf{R}} - (\mathbf{R} \cdot \dot{\mathbf{R}}) \ddot{\mathbf{R}}], \end{aligned}$$

where $\mathbf{h} = \mathbf{R} \times \dot{\mathbf{R}}$.

$$e_{21}(t) = \frac{m_0 m_1}{(m_0 + m_1)^{\frac{4}{3}} M^{\frac{2}{3}} X^{\frac{4}{3}}} \left[\frac{3}{4} \cos l_2 + e_1^2 \left(\frac{33}{16} \cos l_2 + \frac{35}{16} \cos 3l_2 \right) + \frac{P_1(t)}{X} \right] + C_1 \cos K_1 t + C_2 \sin K_1 t + \frac{K_2}{K_1 - K_3} \cos K_3 t$$

X – period ratio

l_2 planetary mean anomaly

$$e_{22}(t) = \frac{m_0 m_1}{(m_0 + m_1)^{\frac{4}{3}} M^{\frac{2}{3}} X^{\frac{4}{3}}} \left[\frac{3}{4} \sin l_2 + e_1^2 \left(\frac{3}{16} \sin l_2 + \frac{35}{16} \sin 3l_2 \right) + \frac{P_2(t)}{X} \right] + C_1 \sin K_1 t - C_2 \cos K_1 t + \frac{K_2}{K_1 - K_3} \sin K_3 t,$$

where

$$C_1 = -\frac{m_0 m_1}{(m_0 + m_1)^{\frac{4}{3}} M^{\frac{2}{3}} X^{\frac{4}{3}}} \left[\frac{3}{4} \cos l_{20} + e_1^2 \left(\frac{33}{16} \cos l_{20} + \frac{35}{16} \cos 3l_{20} \right) + \frac{P_1(t_0)}{X} \right] - \frac{K_2}{K_1 - K_3}$$

$$C_2 = \frac{m_0 m_1}{(m_0 + m_1)^{\frac{4}{3}} M^{\frac{2}{3}} X^{\frac{4}{3}}} \left[\frac{3}{4} \sin l_{20} + e_1^2 \left(\frac{3}{16} \sin l_{20} + \frac{35}{16} \sin 3l_{20} \right) + \frac{P_2(t_0)}{X} \right].$$

$$K_1 = \frac{3 \sqrt{GM} m_0 m_1 a_{1l}^2}{8 (m_0 + m_1)^2 a_{2l}^{\frac{7}{2}}} (2 + 3e_{1l}^2)$$

$$K_2 = \frac{15 \sqrt{GM} m_0 m_1 (m_0 - m_1) a_{1l}^3}{64 (m_0 + m_1)^3 a_{2l}^{\frac{9}{2}}} e_{1l} (4 + 3e_{1l}^2)$$

$$K_3 = \frac{3 \sqrt{G} m_2 a_{1l}^{\frac{3}{2}} \sqrt{1 - e_{1l}^2}}{4 (m_0 + m_1)^{\frac{1}{2}} a_{2l}^3}$$

$$\begin{aligned}
 P_1(t) = & \frac{21}{32}(1 - \sqrt{1 - e_1^2}) \cos(2E_1 + 3l_2) + \frac{3}{32}(1 - \sqrt{1 - e_1^2}) \cos(2E_1 + l_2) - \frac{3}{32}(1 + \\
 & + \sqrt{1 - e_1^2}) \cos(2E_1 - l_2) - \frac{21}{32}(1 + \sqrt{1 - e_1^2}) \cos(2E_1 - 3l_2) + e_1 \left[-\frac{21}{96}(1 - \sqrt{1 - e_1^2}) \cos(3E_1 + 3l_2) - \frac{3}{96}(1 - \sqrt{1 - e_1^2}) \cos(3E_1 + l_2) + \frac{3}{32}(13 + \right. \\
 & + 5\sqrt{1 - e_1^2}) \cos(E_1 - l_2) + \frac{3}{96}(1 + \sqrt{1 - e_1^2}) \cos(3E_1 - l_2) - \frac{105}{32}(1 - \sqrt{1 - e_1^2}) \cos(E_1 + 3l_2) + \frac{21}{96}(1 + \sqrt{1 - e_1^2}) \cos(3E_1 - 3l_2) - \frac{3}{32}(13 - \\
 & - 5\sqrt{1 - e_1^2}) \cos(E_1 + l_2) + \left. \frac{105}{32}(1 + \sqrt{1 - e_1^2}) \cos(E_1 - 3l_2) \right] + e_1^2 \left[\frac{3}{32} \left(\frac{7}{2} - \sqrt{1 - e_1^2} \cos(2E_1 + l_2) + \frac{21}{32} \left(\frac{1}{2} - \sqrt{1 - e_1^2} \cos(2E_1 + 3l_2) - \frac{3}{32} \left(\frac{7}{2} + \right. \right. \right. \right. \\
 & + \sqrt{1 - e_1^2}) \cos(2E_1 - l_2) - \left. \left. \frac{21}{32} \left(\frac{1}{2} + \sqrt{1 - e_1^2} \cos(2E_1 - 3l_2) \right) \right) + e_1^3 \left[-\frac{1}{64} \times \right. \\
 & \times \cos(3E_1 + l_2) - \frac{7}{64} \cos(3E_1 - 3l_2) + \frac{3}{64} \cos(E_1 - l_2) + \frac{7}{64} \cos(3E_1 + 3l_2) + \\
 & + \frac{21}{64} \cos(E_1 + 3l_2) - \frac{3}{64} \cos(E_1 + l_2) + \frac{1}{64} \cos(3E_1 - l_2) - \\
 & \left. \left. \left. - \frac{21}{64} \cos(E_1 - 3l_2) \right] \right]
 \end{aligned}$$

$$\begin{aligned}
 P_2(t) = & \frac{21}{32}(1 - \sqrt{1 - e_1^2}) \sin(2E_1 + 3l_2) - \frac{3}{32}(1 - \sqrt{1 - e_1^2}) \sin(2E_1 + l_2) - \frac{3}{32}(1 + \\
 & + \sqrt{1 - e_1^2}) \sin(2E_1 - l_2) + \frac{21}{32}(1 + \sqrt{1 - e_1^2}) \sin(2E_1 - 3l_2) + e_1 \left[-\frac{7}{32}(1 - \sqrt{1 - e_1^2}) \sin(3E_1 + 3l_2) + \frac{1}{32}(1 - \sqrt{1 - e_1^2}) \sin(3E_1 + l_2) - \frac{3}{32}(3 - \right. \\
 & - 5\sqrt{1 - e_1^2}) \sin(E_1 - l_2) + \frac{1}{32}(1 + \sqrt{1 - e_1^2}) \sin(3E_1 - l_2) - \frac{105}{32}(1 - \sqrt{1 - e_1^2}) \sin(E_1 + 3l_2) - \frac{7}{32}(1 + \sqrt{1 - e_1^2}) \sin(3E_1 - 3l_2) - \frac{3}{32}(3 + \\
 & + 5\sqrt{1 - e_1^2}) \sin(E_1 + l_2) - \left. \frac{105}{32}(1 + \sqrt{1 - e_1^2}) \sin(E_1 - 3l_2) \right] + e_1^2 \left[\frac{3}{32} \left(\frac{5}{2} + \sqrt{1 - e_1^2} \sin(2E_1 + l_2) + \frac{21}{32} \left(\frac{1}{2} - \sqrt{1 - e_1^2} \sin(2E_1 + 3l_2) + \frac{3}{32} \left(\frac{5}{2} - \right. \right. \right. \right. \\
 & - \sqrt{1 - e_1^2}) \sin(2E_1 - l_2) + \left. \left. \frac{21}{32} \left(\frac{1}{2} + \sqrt{1 - e_1^2} \sin(2E_1 - 3l_2) \right) \right) + e_1^3 \left[-\frac{3}{64} \times \right. \\
 & \times \sin(3E_1 + l_2) + \frac{7}{64} \sin(3E_1 - 3l_2) - \frac{9}{64} \sin(E_1 - l_2) + \frac{7}{64} \sin(3E_1 + 3l_2) + \\
 & + \frac{21}{64} \sin(E_1 + 3l_2) - \frac{9}{64} \sin(E_1 + l_2) - \frac{3}{64} \sin(3E_1 - l_2) + \\
 & \left. \left. \left. + \frac{21}{64} \sin(E_1 - 3l_2) \right] \right]
 \end{aligned}$$

Maximum and averaged squared eccentricity

$$\begin{aligned}
 e_2^{max} &= e_{2sm}^{max} + e_{2l}^{max} \\
 &= \frac{m_0 m_1}{(m_0 + m_1)^{\frac{4}{3}} M^{\frac{2}{3}} X^{\frac{4}{3}}} \left[\frac{3}{2} + \frac{17}{2} e_1^2 + \frac{1}{X} \left(3 + 19e_1 + \frac{21}{8} e_1^2 - \frac{3}{2} e_1^3 \right) \right] \\
 &\quad + \frac{2K_2}{K_1 - K_3}.
 \end{aligned}$$

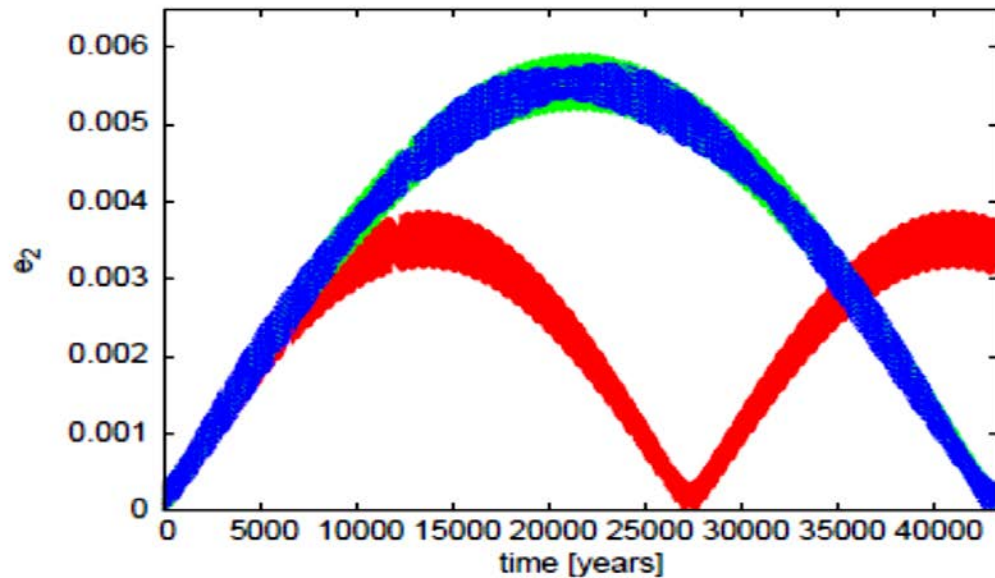
$$\begin{aligned}
 \langle e_2^2 \rangle &= \langle e_{21}^2(t) + e_{22}^2(t) \rangle = \frac{m_0^2 m_1^2}{(m_0 + m_1)^{\frac{8}{3}} M^{\frac{4}{3}} X^{\frac{8}{3}}} \left[\frac{9}{8} + \frac{27}{8} e_1^2 + \frac{887}{64} e_1^4 - \right. \\
 &\quad \left. - \frac{975}{64} \frac{1}{X} e_1^4 \sqrt{1 - e_1^2} + \frac{1}{X^2} \left(\frac{225}{64} + \frac{6619}{64} e_1^2 - \frac{26309}{512} e_1^4 - \frac{393}{64} e_1^6 \right) \right] + \\
 &\quad + 2 \left(\frac{K_2}{K_1 - K_3} \right)^2.
 \end{aligned}$$

Post-Newtonian correction

$$\frac{dg_{1PN}}{dt} = \frac{3\mathcal{G}^{\frac{3}{2}}(m_0 + m_1)^{\frac{3}{2}}}{c^2 a_{1l}^{\frac{5}{2}}(1 - e_{1l}^2)}$$

$$K_{3PN} = \frac{3\mathcal{G}^{\frac{1}{2}}m_2 a_{1l}^{\frac{3}{2}}(1 - e_{1l}^2)^{\frac{1}{2}}}{4(m_0 + m_1)^{\frac{1}{2}}a_{2l}^3} + \frac{3\mathcal{G}^{\frac{3}{2}}(m_0 + m_1)^{\frac{3}{2}}}{c^2 a_{1l}^{\frac{5}{2}}(1 - e_{1l}^2)}$$

Red – non-relativistic full equations of motion
Green – relativistic full equations of motion
Blue – analytical solution



$$m_0 = 5 M_{\odot}, \quad m_1 = 4 M_{\odot}, \quad m_2 = 1 M_J$$

$$a_1 = 0.2 \text{ au}, \quad a_2 = 6.84 \text{ au}, \quad e_1 = 0.5$$

$$E_{10} = 0^{\circ}, \quad \ell_{20} = 90^{\circ}$$

Analytic description of the system's evolution

Binary orbit

$$\varpi_1 \approx g_1 = K_{3\text{PN}}t,$$

$$l_1 = n_1t + l_{10},$$

$$r \cos f_1 = a_1(\cos E_1 - e_1),$$

$$r \sin f_1 = a_1(1 - e_1^2)^{1/2} \sin E_1,$$

$$n_1 = \mathcal{G}^{1/2}(m_0 + m_1)^{1/2}a_1^{-3/2}$$

$$r = W_1(r \cos f_1, r \sin f_1)^T$$

$$W_i = \begin{pmatrix} \cos \varpi_i & -\sin \varpi_i \\ \sin \varpi_i & \cos \varpi_i \end{pmatrix}, \quad i = 1, 2$$

$$\dot{a}_1 = 0, \dot{a}_2 = 0, \dot{e}_1 = 0$$

Planetary orbit

$$\varpi_2 = \arctan(e_{22}/e_{21}),$$

$$l_2 = n_2t + \varpi_2 + l_{20},$$

$$e_2(t) = |e_2|(t),$$

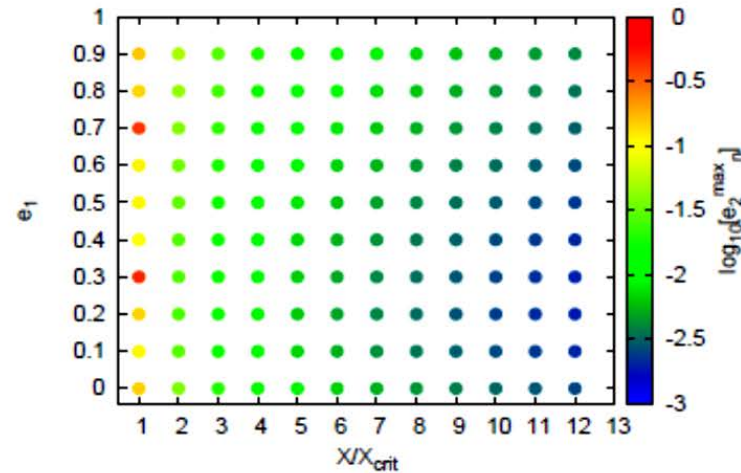
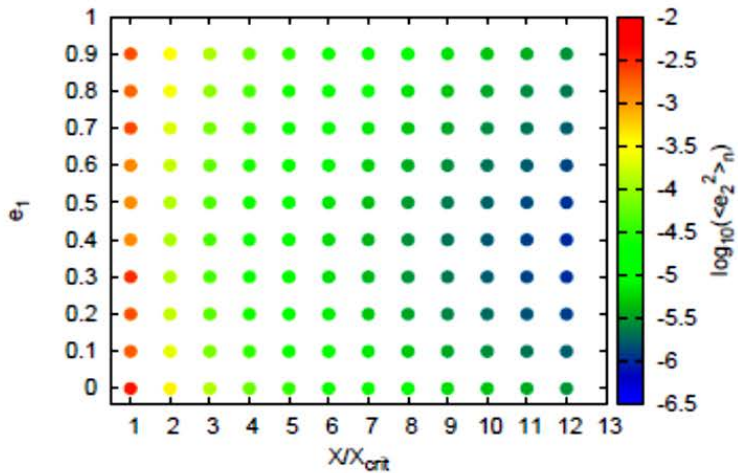
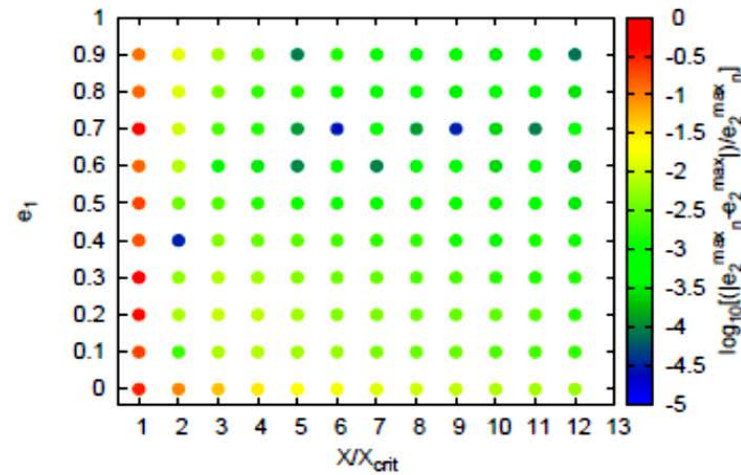
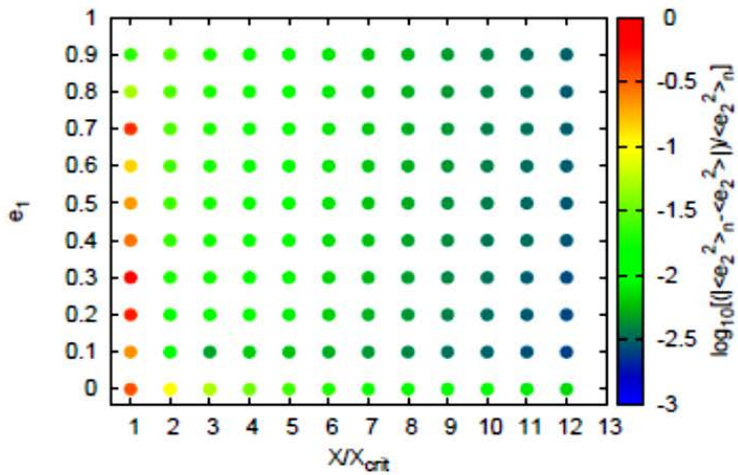
$$R \cos f_2 = a_2(\cos E_2 - e_2),$$

$$R \sin f_2 = a_2(1 - e_2^2)^{1/2} \sin E_2$$

$$n_2 = (\mathcal{GM})^{1/2}a_2^{-3/2}$$

$$R = W_2(R \cos f_2, R \sin f_2)^T$$

Numerical Testing



$$m_1/(m_0+m_1)=0.3$$

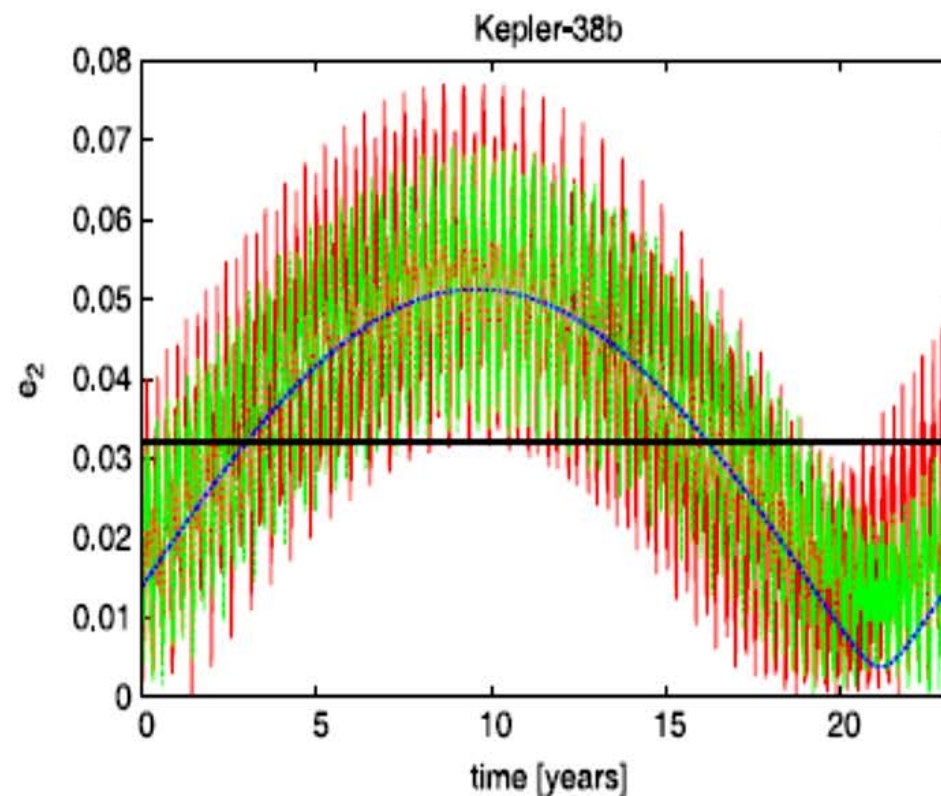
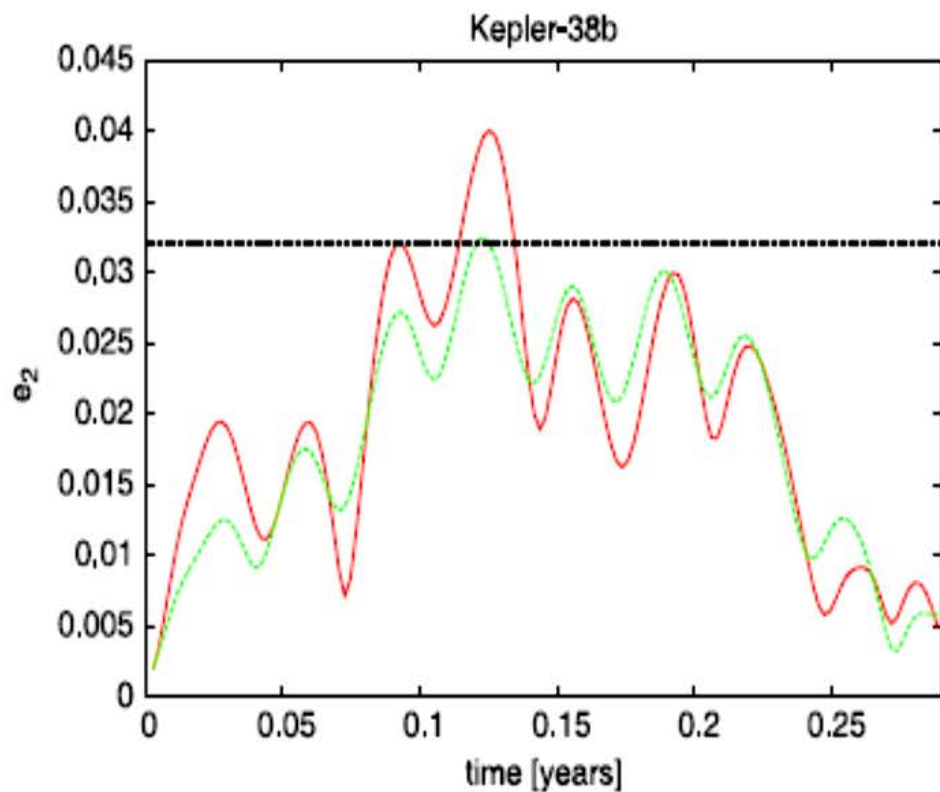
$$m_2/(m_0+m_1)=0.001$$

X_{crit} is the critical period ratio based on Holman & Wiegert (1999).

Integration time= 1 analytical secular period

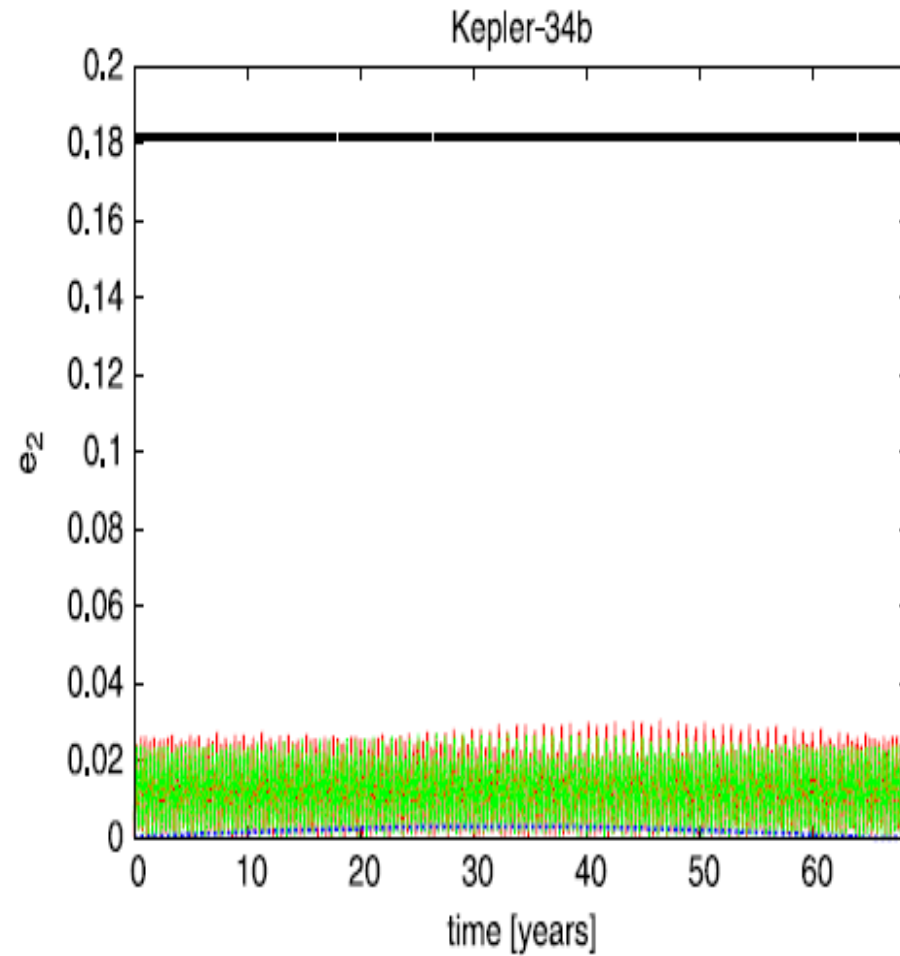
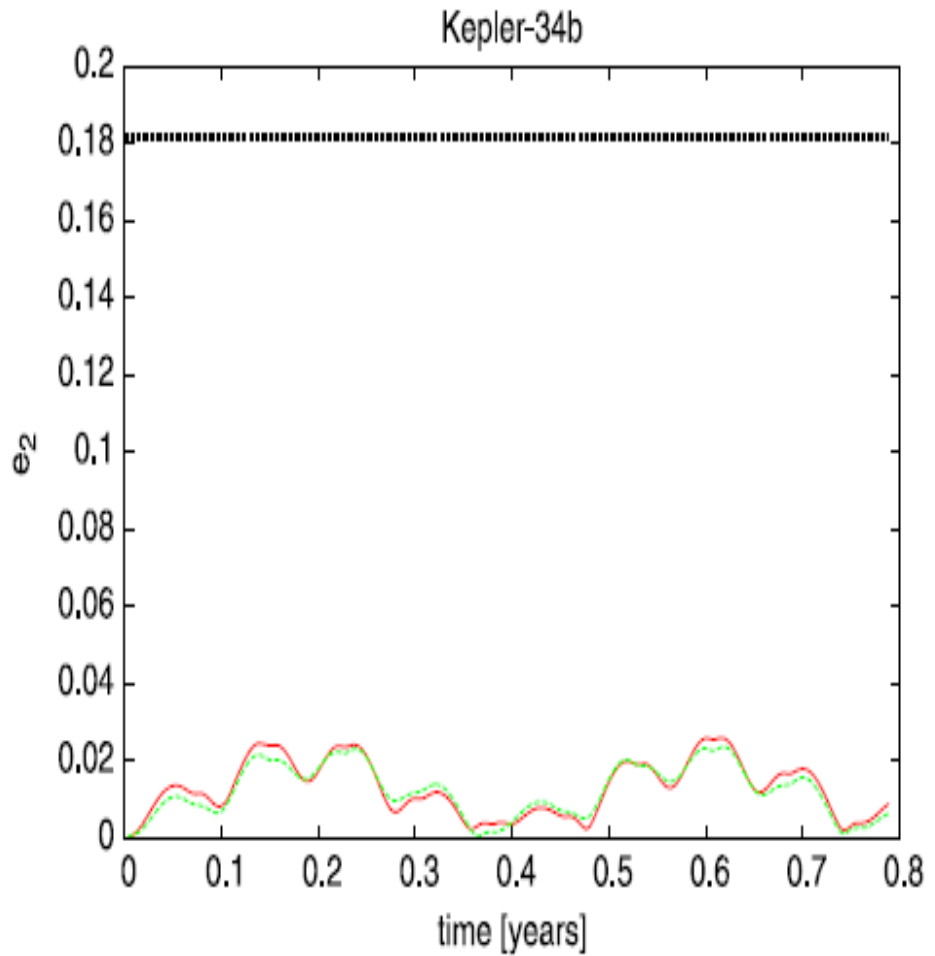
APPLICATION TO REAL SYSTEMS

Kepler-16, Kepler-34, Kepler-35, Kepler-38, Kepler-64, Kepler-413



Red – Numerical solution
Green – Analytical solution
Blue – Analytical secular solution
Black – Observed value

$m_0 = 0.949 M_\odot$
 $m_1 = 0.249 M_\odot$
 $m_2 = 0.384 M_J$
 $a_1 = 0.1469 \text{ au}$
 $a_2 = 0.4644 \text{ au}$
 $e_1 = 0.1032$
 $E_{10} = 0^\circ$
 $\varpi_1 = 0^\circ$
 $\iota_{20} = 90^\circ$

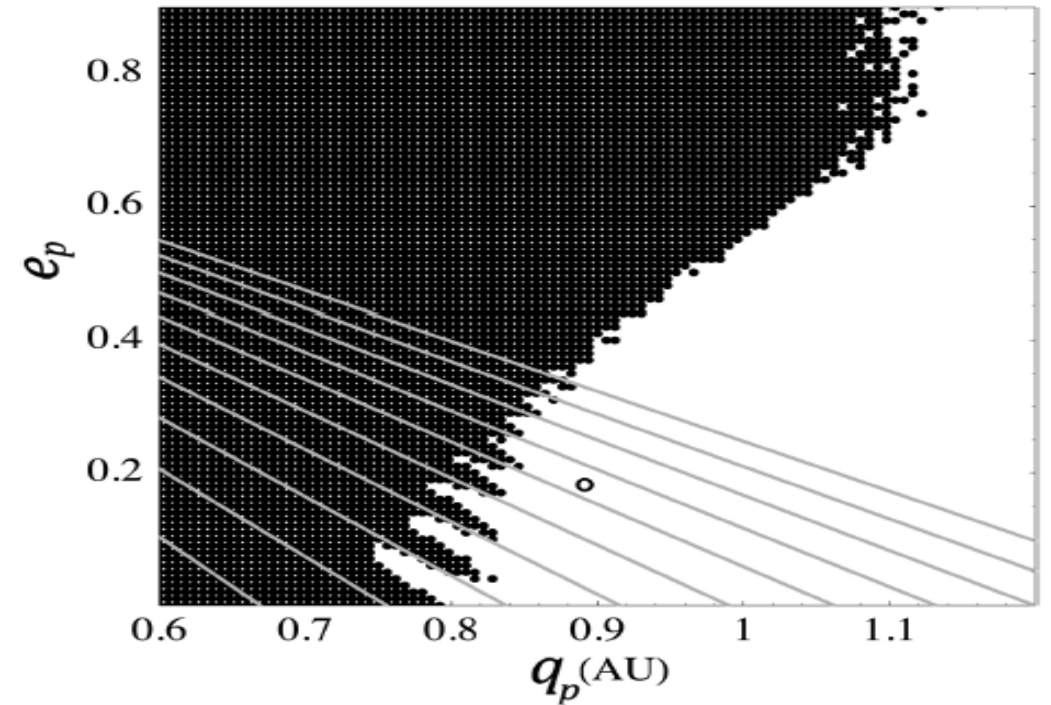
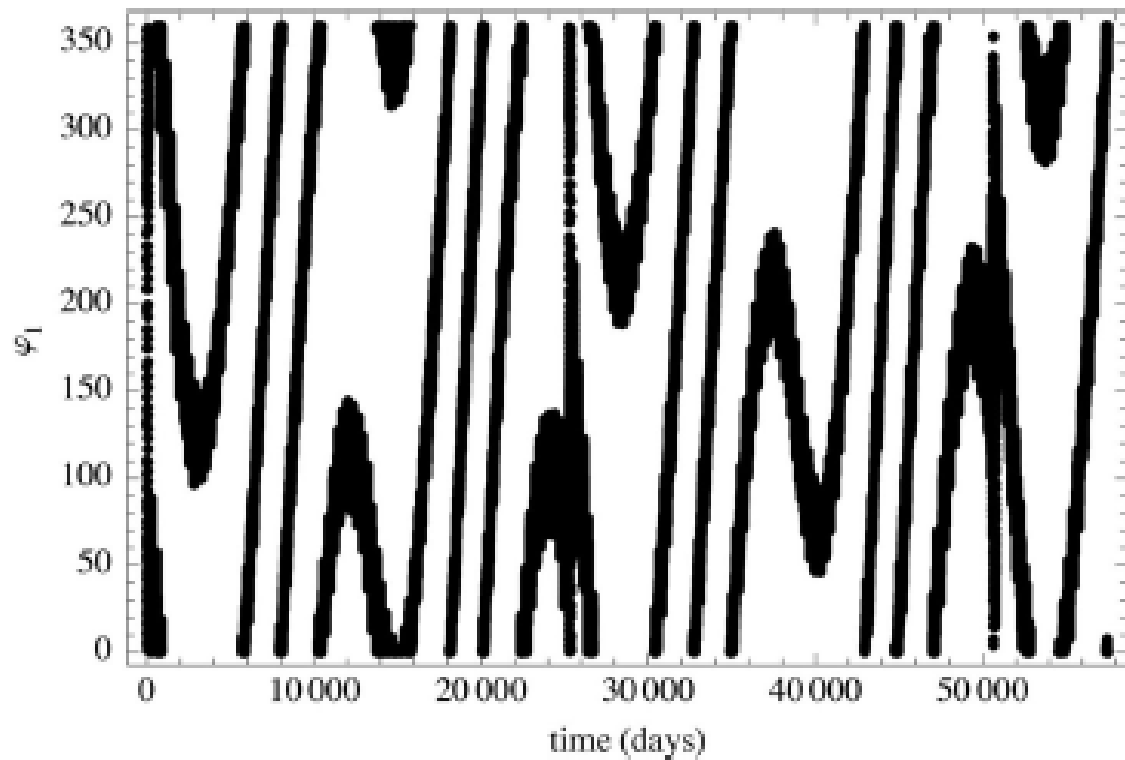


Red – Numerical solution
Green – Analytical solution
Blue – Analytical secular solution
Black – Observed value

$m_0=1.0479 M_\odot$
 $m_1=1.0208 M_\odot$
 $m_2=0.220 M_J$
 $a_1=0.22882 \text{ au}$
 $a_2=1.0896 \text{ au}$
 $e_1=0.52087$
 $E_{10}=0^\circ$
 $\varpi_1=0^\circ$
 $\zeta_{20}=90^\circ$

Chavez et al. (2015)

$$\varphi_1 = \lambda_b - 10\lambda_p + 9\varpi_p$$



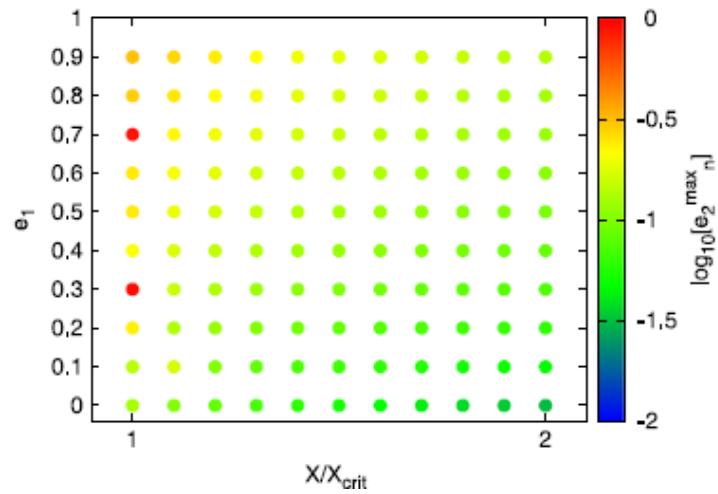
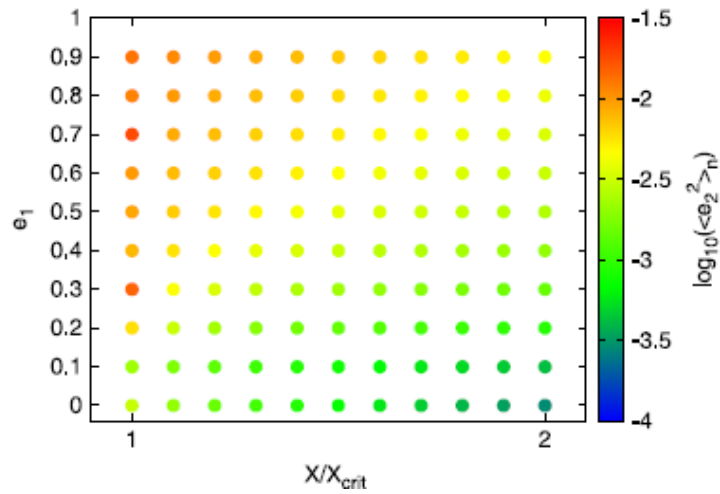
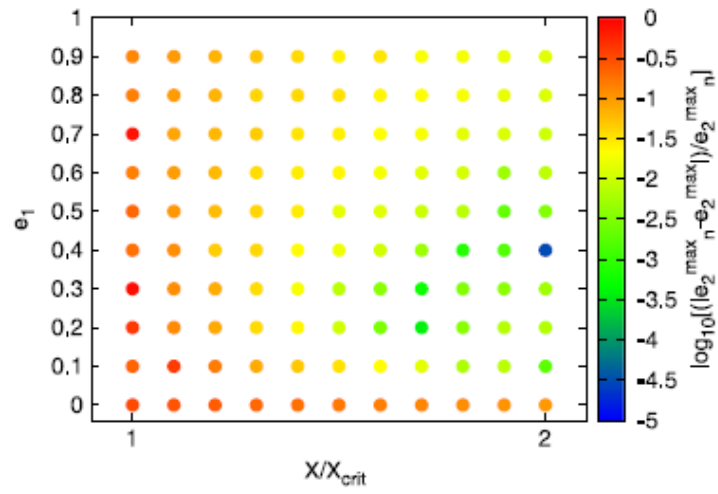
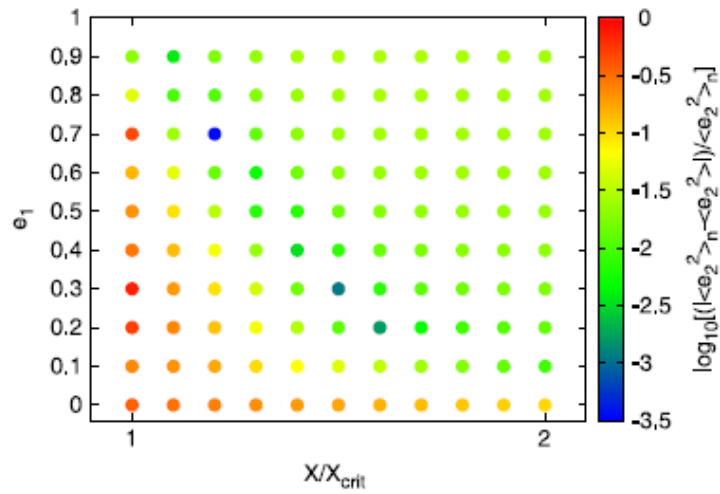
resonance trapping? (Kley & Haghighipour 2014) + 2nd planet? (Kley & Haghighipour 2015)

More details can be found in: Georgakarakos, N., Eggl, S. (2015), *ApJ*, 802, 94

THANK YOU FOR YOUR ATTENTION

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Thessaloniki, June 28 – July 2, 2015



$$m_1/(m_0+m_1)=0.3$$

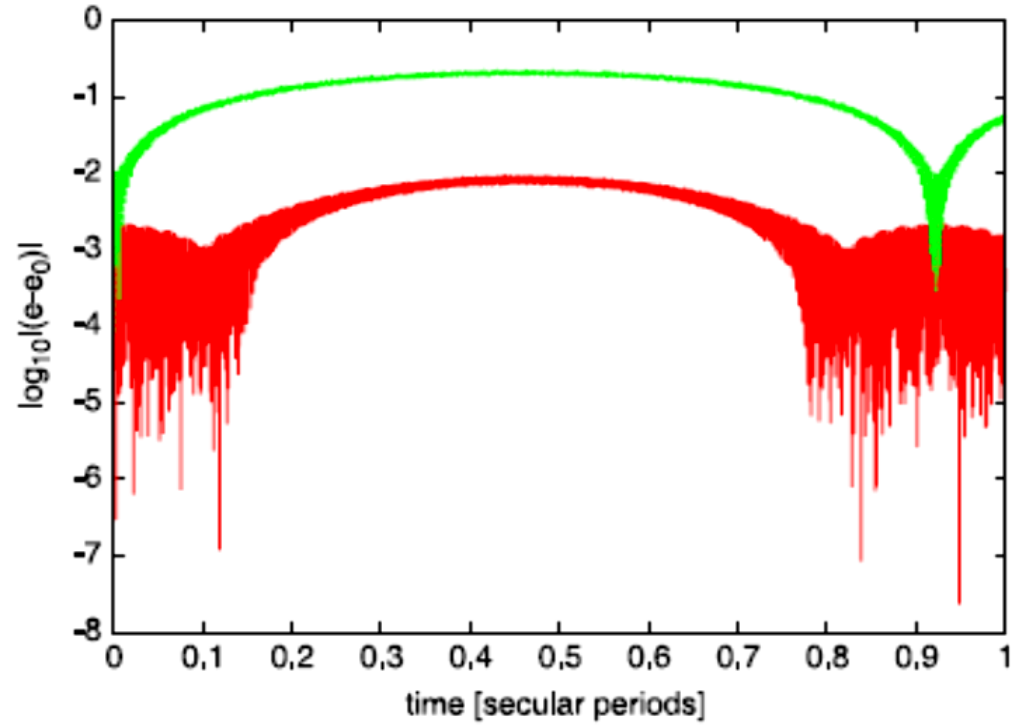
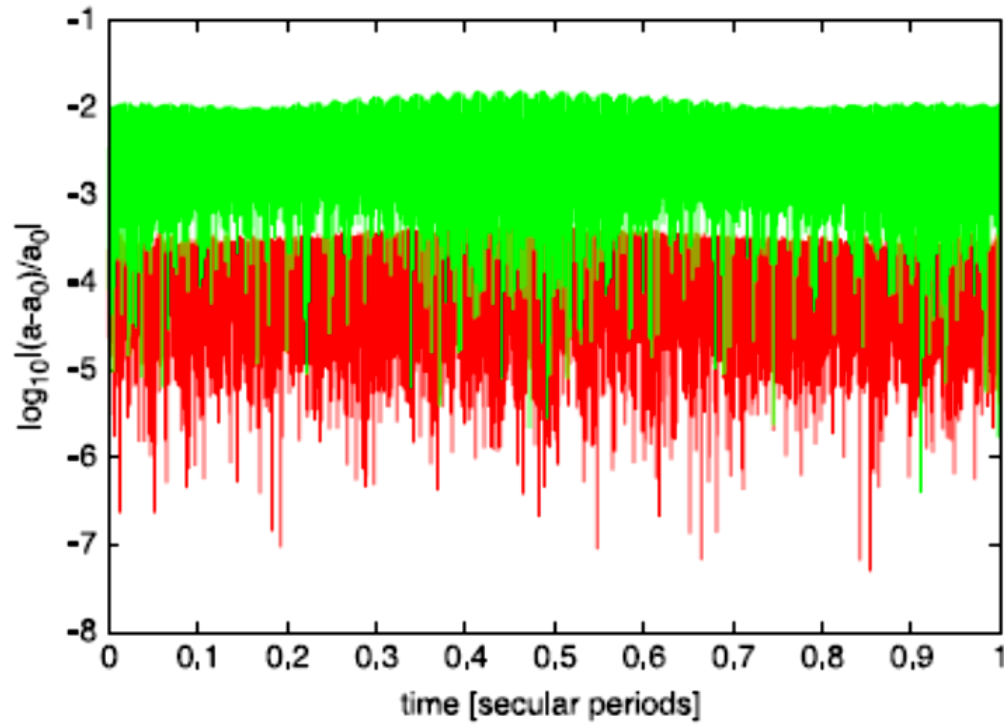
$$m_2/(m_0+m_1)=0.001$$

X_{crit} is the critical period ratio based on Holman & Wiegert (1999).

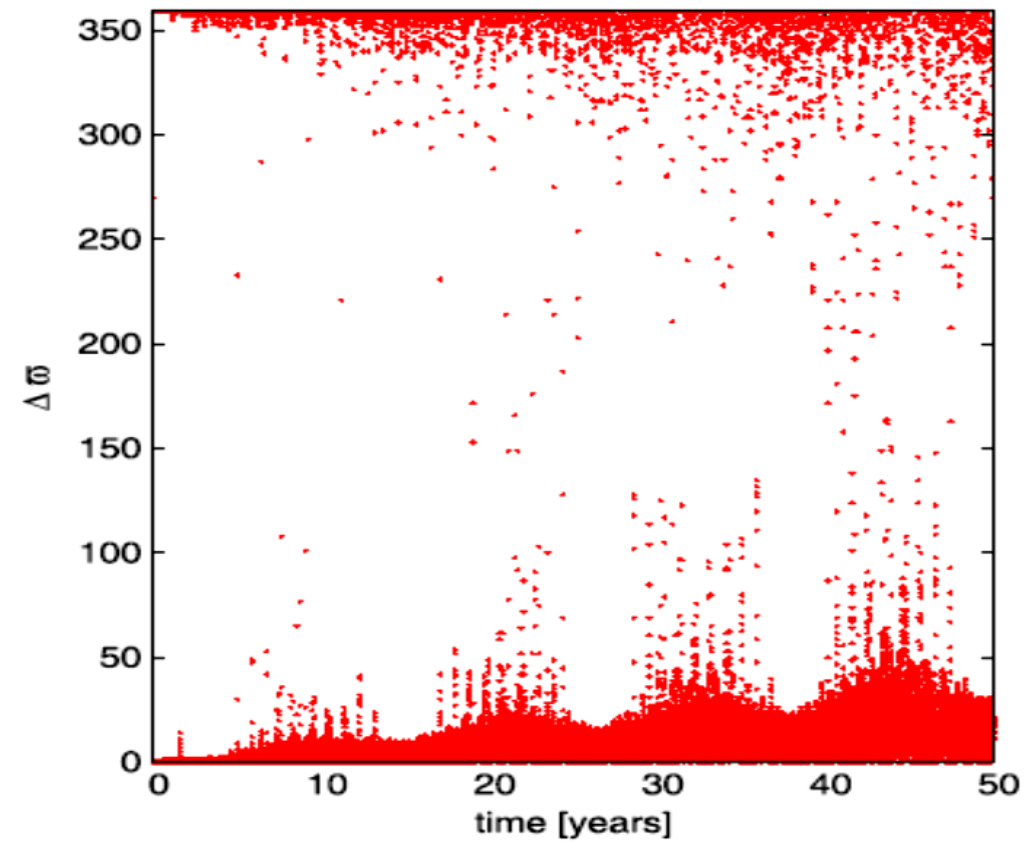
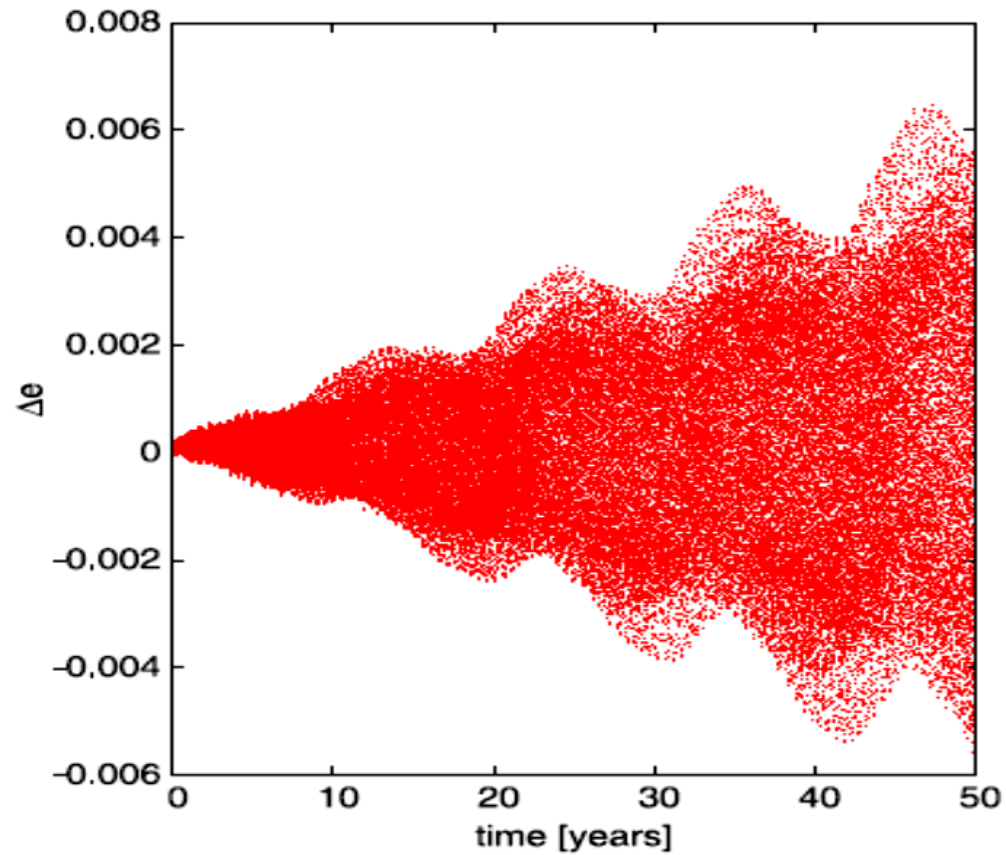
Integration time= 1 analytical secular period

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Kepler-413



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